## CRITICAL ITEMS LIST

PROJECT: SRMS (-5 NCIU INSTALLED)
ASS'Y NOMENCLATURE: SROULDER

SYSTEM: MECHANICAL ARM SUBSYSTEM ASS'Y P/N: 51140J1219

SHEET: \_ NAME OTY & DRAWING REF. **FHEA FHEA FAILURE HODE** FAILURE EFFECT HOWR / FUNC. RATIONALE FOR ACCEPTANCE REf. REV. AND ALC: 2/1R DESIGNATION CAUSE END ITEM CRITICALITY SCREENS: A-PASS, B-PASS, C-PASS 4615 0 SHOULDER MODE: LOSS OF 28V TO DESIGN FEATURES FUSING. 45 LOSS OF JPC. ALL DATA COMMUNICATION SHOULDER/ PRIME CHANNEL **ELBOW JPC** AND PRIME FUSES USED IN THE SHOULDER FUSE PLUG ASSEMBLIES ARE OF THE FUSES. 16 FUSE. CHANNEL DESIGN DEFINED BY MSFC SPECIFICATION 40M38259. FOR SRMS BACK-UP FUNCTIONS WILL APPLICATION, DESIGN AND PROCESS IMPROVEMENTS HAVE BEEN CHANNEL NEGOTIATED WITH, AND IMPLEMENTED BY, THE HANUFACTURER. STOP. FUSES. AUTOBRAKES ARE CAUSE(S): THESE INCLUDE: WIRING APPLIED. ARM SCHEMATIC MÉCHANICAL COMES TO REST. - IMPROVED ATTACHMENT OF END CAPS. 51140E316 SHOCK, VIBRATION - CONTROL OF FUSE ELEMENT LENGTH AND DISPOSITION WITHIN THE LOSS OF REVISION C. COMPUTER MATERIALS SUPPORTED - CONTROL SOLDERING BETWEEN FUSE ELEMENT AND THE END CAPS. (FUSE 15). MODES. LOSS OF LIMPING DURING PRIOR TO ASSEMBLY IN THE FUSE PLUG ASSEMBLY, A CONNECT PIN 1S SOLDERED TO EACH OF THE FUSE LEAD WIRES. THIS PROCESS IS END EFFECTOR CAPTURE. IF WRIST JPC: END CONTROLLED BY ESTABLISHED PROCEDURES WHICH INCLUDE THE REQUIREMENT OF A "METERED" QUALITY OF SOLDER FOR EACH SOLDER EFFECTOR AUTO JOINT. THE FUSE BODY AND LEAD WIRES ARE SLEEVED TO PRECLUDE DRIVE MODE MAY SHORT CIRCUITS. EACH FUSE AND ALL SOLDERED JOINTS ARE SUBJECTED TO RADIOGRAPHIC INSPECTION. NOT FUNCTION CORRECTLY. THE FUSE PLUG ASSEMBLY INCLUDES AN ALLMENUM POTTING SHELL. FOLLOWING INTEGRATION OF THE FUSES, THE CONNECTOR WORST CASE ASSEMBLY IS POTTED USING A SEMI-RESILIENT (RTV) COMPOUND. THE POTTING MEDIUM PROVIDES GOOD HEAT TRANSFER AND ENSURES UNEXPECTED MECHANICAL STABILITY OF THE INDIVIDUAL FUSES. MOTION FREE NO REDUNDANCY IS PROVIDED FOR THE SHOULDER/ELBOW JPC FUSING. JOINTS. AUTO BRAKES. REDUNDANT PATHS REMAINING 1) AUTOBRAKES (TO SAFE THE SYSTEM). 2) BACK-UP DRIVE AND END EFFECTOR NANUAL DRIVE MODES (TO SECURE ORBITER).

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PREPARED 8Y:

MFWG

SUPERCEDING DATE: NONE

RMS/MECH - 361

DATE: 11 JUL 91

CIL REV: \_O

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## CRITICAL ITEMS LIST

PROJECT: SRMS (-5 MCIU INSTALLED)
ASS'Y NOMENCLATURE: SHOULDER

SYSTEM: MECHANICAL ARM SUBSYSTEM ASS'Y P/N: 51140J1219

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FMEA Ref.	FMEA REV.	HAME, QTY, & Drawing Ref.	FAILURE MODE	FAILURE EFFECT ON	HDWR / FUNC. RATIONALE FOR ACCEPTANCE 2/1R
		DESIGNATION	CAUSE	END ITEM	CRITICALITY SCREENS: A-PASS, B-PASS, C-PASS
4615	O	SHOULDER FUSING. 45 PRIME CHANNEL FUSES. 16 BACK-UP CHANNEL FUSES. VIRING SCHEMATIC 51140E316 REVISION C.	MODE: LOSS OF SHOULDER/ ELBOW JPC FUSE.  CAUSE(S): (1) MECHANICAL SHOCK, VIBRATION MATERIALS (FUSE 15).	LOSS OF 28Y TO JPC. ALL DATA COMMUNICATION AND PRIME CHANNEL FUNCTIONS WILL STOP. AUTOBRAKES ARE APPLIED. ARM COMES TO REST. LOSS OF L	GA/INSPECTIONS  FUSES ARE PROCURED AS A EEE PART TO SPAR SPECIFICATION SPAR-SG459/023, WHICH INCOMPORATES SPECIFICATION MSFC40M38259 AS REQUIRED BY SPAR-RMS-PA.003. QUALIFICATION, ACCEPTANCE TESTING AND RELIABILITY LIFE TESTING OF FUSE PLUG ASSEMBLIES WAS PERFORMED TO THE REQUIREMENTS OF THE SPAR-RMS-TP.952.  EEF PARTS IMSPECTION IS PERFORMED AS REQUIRED BY SPAR-RMS-PA.003. EACH EEE PART IS QUALIFIED AT THE PART LEVEL TO THE REQUIREMENTS OF THE APPLICABLE SPECIFICATION. ALL EEE PARTS ARE 100X SCREENED AND BURNED IN, AS A MINIMUM, AS REQUIRED BY SPAR RMS-PA.003, BY THE SUPPLIER. ADDITIONALLY, EEE PARTS ANE 100X RE-SCREENED IN ACCORDANCE WITH REQUIREMENTS BY AN INDEPENDENT SPAR APPROVED TESTING FACILITY. DPA IS PERFORMED AS REQUIRED BY PA.003 ON A RANDOMLY SELECTED 5X OF PARTS, MIXIMUM 5 PIECES, MINIMUM 3 PIECES FOR EACH LOT MUMBER/DATE CODE OF PARTS RECEIVED.  WIRE IS PROCURED TO SPECIFICATION MILL-22759 OR MIL81381 AND INSPECTED AND TESTED TO NASA JSCHBORD STANDARD HUMBER 95A.  RECEIVING INSPECTION VERIFIES THAT ALL PARTS RECEIVED ARE AS IDENTIFIED IN THE PROCURENENT DOCUMENTS, THAT NO PHYSICAL DAMAGE HAS OCCURRED TO PARTS DURING SHIPHENT, THAT THE RECEIVING DOCUMENTS PROVIDE ADEQUATE TRACEABILITY INFORMATION AND SCREENING DATA CLEARLY IDENTIFIES ACCEPTABLE PARTS.  PARTS ARE INSPECTED INROUGHOUT MANUFACTURE AND ASSEMBLY AS APPROPRIATE TO THE MANUFACTURING STAGE COMPLETED. THESE INSPECTIONS INCLUDE,  COMPONENT MOUNTING INSPECTION FOR CORRECT SOLDERING, WIRE LOOPING, STRAPPING, ETC. OPERATORS AND INSPECTION TO CHECK FOR POROSITY AND INTERNAL FLAMS.  PRE AND POST POTTING THE SOLDERED CONTACTS OF THE FUSE ASSEMBLY ARE SUBJECTED TO RADIOGRAPHIC INSPECTION TO CHECK FOR POROSITY AND INTERNAL FLAMS.  PRE AND POST POTTING INSPECTION STO SPAR-ITP 257 VERIFY VOLTAGE BOOP (RESISTANCE) AT HIGH AND LOW IEMPERATURE (-38) DECREES C AND +118 DECREES C) (SPAR/GOVERNMENT REP.  MALDATORY INSPECTION POINT).  FUSES ARE ACCEPTANCE TESTED TO SPAR-ITP 257 WELLEY  WOLTAGE BROW INSPECTION TO SPAR-ITP 257 WELLEY  WOLTAGE BROW INSPECTION

S040237A ATTACHMENT -PAGE 235 OF 471 PROJECT: SRMS (-5 MCIU INSTALLED)
ASS'Y NOMENCLATURE: SHOULDER

SYSTEM: MECHANICAL ARM SUBSYSTEM ASS'Y P/N: 5114011219

SHEET: \_\_4

FMEA FMEA REV.	MAME, QTY, L DRAWING REF. DESIGNATION	FAILURE MODE AND CAUSE	FAILURE EFFECT ON END ITEM	HDWR / FUNC. RATIONALE FOR ACCEPTANCE 2/1R CRITICALITY SCREENS: A-PASS, B-PASS, C-PASS
4615 0	SHOULDER FUSING. 45 PRIME CHANNEL FUSES. 16 BACK-UP CHANNEL FUSES. WIRING SCHEMATIC 511406316 REVISION C.	MODE: LOSS OF SHOULDER/ ELBOM JPC FUSE.  CAUSE(S): (1) MECHANICAL SHOCK VIBRATION NATERIALS (FUSE 15).	LOSS OF 28V TO JPC. ALL DATA COMMUNICATION AND PRIME CHANNEL FUNCTIONS WILL STOP. AUTOBRAKES ARE APPLIED. ARM COMES TO REST. LOSS OF COMPUTER SUPPORTED MODES. LOSS OF LIMPING DURING END EFFECTOR AUTO DRIVE MODE MAY NOT FUNCTION CORRECTLY.  WORST CASE UNEXPECTED MOTION FREE JOINTS. AUTO BRAKES.  REDUNDANT PATHS REMAINING TO SAFE THE SYSTEM).  2) BACK-UP DRIVE MODE STEETOR CORBITER).	A TEST READINESS REVIEW (TRR) WHICH INCLUDES VERIFICATION OF TEST PERSONNEL, TEST DOCUMENTS, TEST EQUIPMENT CALIBRATION/VALIDATION STATUS AND HARDWARE CONFIGURATION IS CONVENCE BY QUALITY ASSURANCE IN CONJUNCTION WITH BEGINEERING, RELIABILITY, CONFIGURATION CONTROL, SUPPLIER AS APPLICABLE, AND THE GOVERNENT REPRESENTATIVE, PRIOR TO THE START OF ANY FORMAL TESTING (ACCEPTANCE OR QUALIFICATION).  JOINT LEVEL ACCEPTANCE TESTING (ATP.) INCLUDES AMBINET, VIBRATION AND THERMAL-VAC TESTING.  (SPAR/GOVERNMENT REP MANDATORY INSPECTION POINT).  SRMS SYSTEMS INTEGRATION, THE INTEGRATION OF MECHANICAL ARM SUBASSENGLIES AND THE FLIGHT CABIN EQUIPMENT TO FORM THE SHIS. INSPECTIONS ARE PERFORMED AT EACH PHASE OF INTEGRATION UNICH INCLUDES GROUNDING CHECKS, THRU WIRING CHECKS, WIRING ROUTING, INTERFACE CONNECTORS FOR BENT OR PUSH BACK CONTACTS ETC.  SRMS SYSTEMS TESTING - STRONGBACK AND FLAT FLOOR AMBIENT PERFORMANCE TEST. (SPAR/GOVERNMENT REP MANDATORY INSPECTION POINT)

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## CRITICAL ITEMS LIST

PROJECT: SRMS (-5 MCIU INSTALLED)
ASS'Y NOMENCLATURE: SHOULDER

SYSTEM: MECHANICAL ARM SUBSYSTEM ASS'Y P/N: 51140J1219

\_\_\_ SHEET: \_\_\_5

SIGULDER PURISE, 55 PRIME CHARGE CHAR	FMEA REF.	FMEA REV.	NAME QTY & DRAWING REF. DESIGNATION	FATLURE MODE AND CAUSE	FAILURE EFFECT ON END TTEM	HDWR / FUNC. RATIONALE FOR ACCEPTANCE 2/1R CRITICALITY SCREENS: A-PASS, B-PASS, C-PASS
	4615	0	FUSING. 45 PRIME CHANNEL FUSES. 16 BACK-UP CHANNEL FUSES. WIRING SCHEMATIC 51140E316	LOSS OF SHOULDER/ ELBOW JPC FUSE. CAUSE(S): (1) MECHAMICAL SHOCK, VIBRATION MATERIALS	JPC. ALL DATA COMMUNICATION AND PRIME CHANNEL FUNCTIONS WILL STOP. AUTOGRAKES ARE APPLIED. ARM COMES TO REST. LOSS OF COMPUTER MODES, LOSS OF LIMPING DURING EMD EFFECTOR CAPTURE. IF WRIST JPC: END EFFECTOR AUTO DRIVE MODE MAY NOT FUNCTION CORRECTLY.  WORST CASE UMEMPECTED MOTION FREE JOINTS. AUTO BRAKES.  REDUNDANT PATHS REMAINING  1) AUTOBRAKES (TO SAFE THE SYSTEM).  2) BACK-UP DRIVE AND END EFFECTOR MANUAL DRIVE MODES (TO SECURE	THERE HAVE BEEN NO FAILURES ASSOCIATED WITH THIS FAILURE NODE ON THE SRMS PROGRAM.

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PREPARED BY:

MFWG

SUPERCEDING DATE: NONE

DATE: 11 JUL 91

CIL REV: 0

PROJECT: SRMS (-5 MCIU INSTALLED)
ASS'Y NOMENCLATURE: SHOULDER

SYSTEM: MECHANICAL ARM SUBSYSTEM
ASS'Y P/N: 5114011219 SHEET: 6

REV. DRAWING REF.	MODE FAILURE EFFECT HOWR / FUNC. RATIONALE FOR ACCEPTANCE
DESIGNATION	END ITEM CRITICALITY SCREENS: A-PASS, B-PASS, C-PASS
	C AND PRIME CHANNEL FUNCTIONS WILL STOP AUTOMATICALLY IF AN UNDETECTED FAILURE OF THE AUTO BRAKES SYSTEM HAS PREVIOUSLY OCCURRED.  CREW ACTION APPLIED. ARM COMES TO REST. LOSS OF CREW CAN ATTEMPT TO COMPLETE THE MISSION IN DIRECT DRIVE. THIS COMPUTER CAN BE ACCOMPLISHED UNDER CERTAIN CONDITIONS WHICH MUST BE SUPPORTED  CREW CAN ACCOMPLISHED UNDER CERTAIN CONDITIONS WHICH MUST BE EVALUATED ON A CASE BY CASE BASIS. APPLY BRAKES TO STOP ARM IF

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